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Magnitude of the resultant aerodynamic force  $=R = (6303)^2 + (835)^2 = 6303.6 \text{ lb} \approx 83.5 \text{ } \div \ddot{=} 0.76^\circ \text{çè } 6303 \div \div \emptyset \theta = \text{Arc Tan } \text{çç} 3V_\infty \sin \theta$  Minimum velocity occurs when  $\sin \theta = 0$ , i.e., when  $\theta = 0^\circ$  and  $180^\circ$ .  $2.10 \text{ V} = V_{\text{min}} = 0$  at  $\theta = 0^\circ$  and  $180^\circ$ , i.e., at its most forward and rearward points.

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<https://downloadlink.org/p/solutions-manual-for-fundamentals-of-aerodynamics-6th-edition-by-anderson/> 2.1 CHAPTER2 If  $p = \text{constant} = P_0$  However, the integral of the surface vector over a closed surface is zero, i.e., Hence, combining Eqs.

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Hence, we can state that the wave drag on the hull is  $D = \frac{1}{2} \rho c^3 V^3 C_D$  where  $c$  is a length scale associated with the hull, say, the maximum width of the hull. Define the drag coefficient as  $C_D = D / \frac{1}{2} \rho V^2 c^2$ . Also, define a similarity parameter called the Froude number,  $Fr = V / \sqrt{g c}$ .

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show all show all steps. For most gases at standard or near standard conditions, the relationship among pressure, density, and temperature is given by the perfect gas equation of state:  $p = \rho R T$ , where  $R$  is the specific gas constant. For air at near standard conditions,  $R = 287 \text{ J} / (\text{kg} \cdot \text{K})$  in the International System of Units and  $R = 1716 \text{ ft} \cdot \text{lb} / (\text{slug} \cdot \text{R})$  in the English Engineering System of Units.

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